

Feedback Control of Manipulators

Equations of motion

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}_j} \right) - \frac{\partial L}{\partial q_j} = Q_j$$



$$\underline{A \ddot{\theta} = b}$$

$$\rightarrow \underline{M(q) \ddot{q} + C(q, \dot{q}) \dot{q} + G(q) = \tau}$$

$M(q)$ - mass matrix

$C(q, \dot{q}) \dot{q}$ - Coriolis acceleration/torque

$G(q)$ - gravity torque

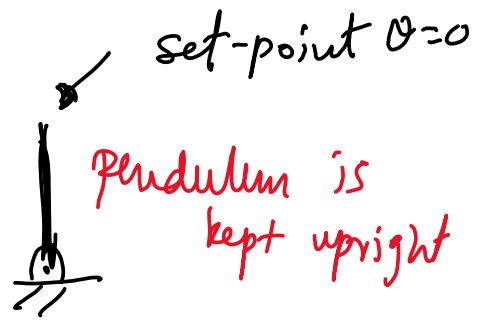
τ - external torque

$$\underline{M(q) \ddot{q}} = \underline{\tau - C(q, \dot{q}) \dot{q} - G(q)}$$

$$\underline{A \ddot{\theta} = b}$$

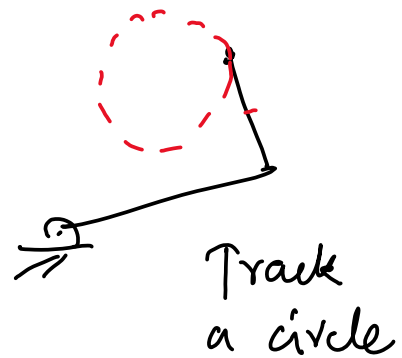
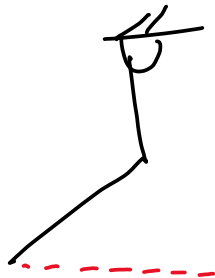
Two objective for control

① Set-point control: e.g.



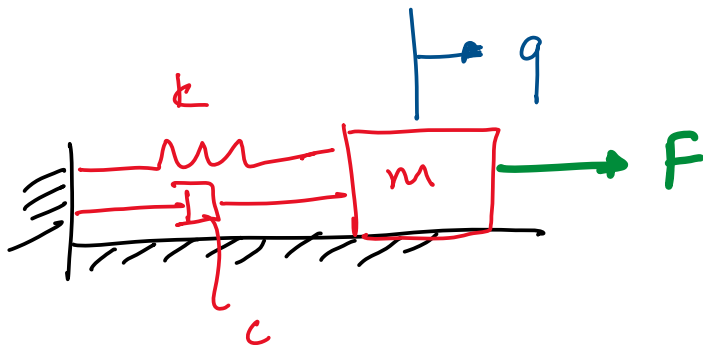
② Trajectory tracking control

Robot
kicking
a ball



$$\underline{M}(q)\ddot{q} + \underline{C}(q, \dot{q})\dot{q} + G(q) = \underline{\tau}$$

$$m\ddot{q} + \underline{c}\dot{q} + \underline{k}q = \underline{F} \sim \text{simple example}$$



Spring mass damper system

$$\ddot{q} + \frac{c}{m} \dot{q} + \frac{k}{m} q = \frac{F}{m} \uparrow \downarrow 0$$

Free Vibrations

$$\omega_n = \sqrt{\frac{k}{m}}$$

$$2 \xi \omega_n = \frac{c}{m}$$

$$\xi = \frac{c}{2\sqrt{mk}}$$

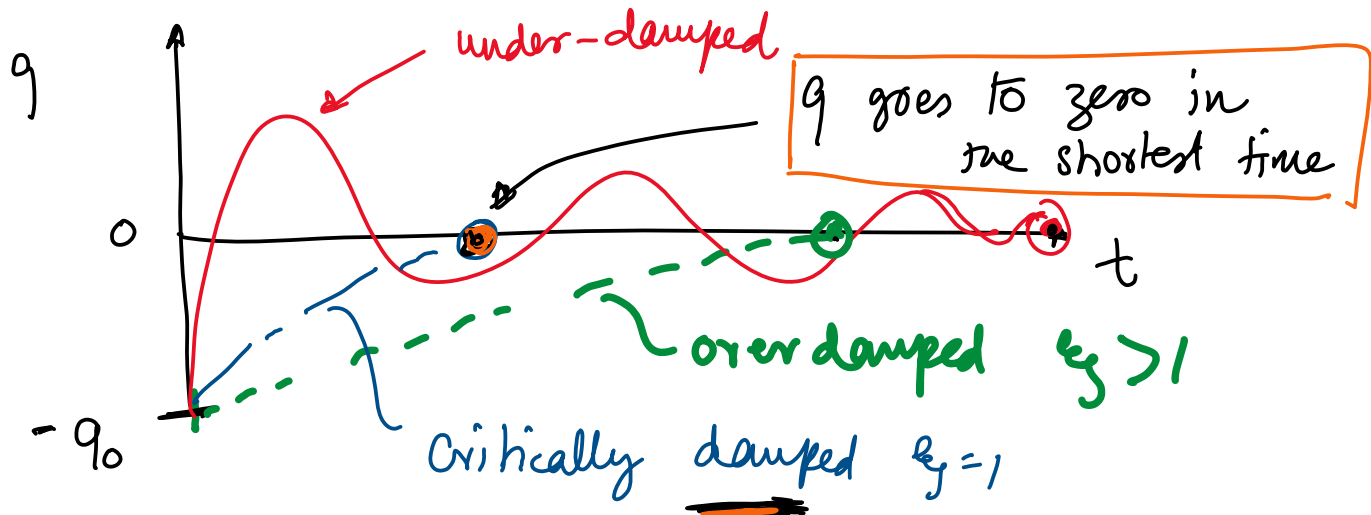
xi

3 cases

① $\xi > 1 \Rightarrow c > 2\sqrt{mk}$ Overdamped

② $\xi = 1 \Rightarrow c = 2\sqrt{mk}$ Critically damped

③ $\xi < 1 \Rightarrow c < 2\sqrt{mk}$ Under damped



$$m\ddot{q} + c\dot{q} + kq = \underline{F}$$

We will design a feedback controller $F(q, \dot{q})$ such that the system is critically damped

$$F = -k_p q - k_d \dot{q}$$

Proportional-derivative controller

$$(-k_p q) \quad (-k_d \dot{q})$$

k_p, k_d — are user-chosen gains.

$$m\ddot{q} + c\dot{q} + kq = \overbrace{-k_p q - k_d \dot{q}}^F$$

$$\rightarrow \underline{m}\ddot{q} + \underline{(c + k_d)}\dot{q} + \underline{(k + k_p)}q = 0$$

Critically damped

$$(c + k_d) = 2\sqrt{m(k + k_p)} \quad \left(\text{or } c = 2\sqrt{mk} \right)$$

Solve for k_d ; keep $k_p = \text{fixed}$

Squaring both sides

$$(c + k_d)^2 = 4m(k + k_p)$$

$$c^2 + 2ck_d + k_d^2 = 4mk + 4mk_p$$

$$k_d^2 + 2ck_d + (c^2 - 4mk - 4mk_p) = 0$$

Solve for k_d

$$k_d = \frac{-2c \pm \sqrt{(2c)^2 - 4(1)(c^2 - 4mk - 4mk_p)}}{2(1)}$$

Take +ve root

$$\rightarrow k_d = -\underline{c} + 2\sqrt{(\underline{k} + \underline{k}_p)\underline{m}}$$

$k_p, k_d \rightarrow$ designer's choice

Extend to a 2D system

$$1D \quad m\ddot{q} + c\dot{q} + kq = F \quad \checkmark$$

2D

$$\left\{ \begin{bmatrix} m_{11} & m_{12} \\ m_{21} & m_{22} \end{bmatrix} \begin{bmatrix} \ddot{q}_1 \\ \ddot{q}_2 \end{bmatrix} + \begin{bmatrix} c_{11} & c_{12} \\ c_{21} & c_{22} \end{bmatrix} \begin{bmatrix} \dot{q}_1 \\ \dot{q}_2 \end{bmatrix} + \begin{bmatrix} k_{11} & k_{12} \\ k_{21} & k_{22} \end{bmatrix} \begin{bmatrix} q_1 \\ q_2 \end{bmatrix} = \begin{bmatrix} F_1 \\ F_2 \end{bmatrix} \right\}$$

$$\rightarrow \underline{F} = -\underline{k}_p q - \underline{k}_d \dot{q}$$

$$\begin{bmatrix} F_1 \\ F_2 \end{bmatrix} = - \underbrace{\begin{bmatrix} k_{p11} & k_{p12} \\ k_{p21} & k_{p22} \end{bmatrix}}_{4 \text{ parameters}} \begin{bmatrix} q_1 \\ q_2 \end{bmatrix} - \underbrace{\begin{bmatrix} k_{d11} & k_{d12} \\ k_{d21} & k_{d22} \end{bmatrix}}_{4 \text{ parameters}} \begin{bmatrix} \dot{q}_1 \\ \dot{q}_2 \end{bmatrix}$$

4 parameters

4 parameters

8 parameters

2 critically damped conditions.

It is difficult to tune 8 parameters with only 2 conditions.

\Rightarrow Feedback linearization

Feedback Linearization / Control partitioning

$$\rightarrow \underline{M} \ddot{q} + \underline{C}(q, \dot{q}) \dot{q} + \underline{G}(q) = \underline{Z} \quad \text{--- ①}$$

$$\rightarrow \underline{Z} = \underline{\hat{M}} (-k_p q - k_d \dot{q}) + \underline{\hat{C}}(q, \dot{q}) \dot{q} + \underline{\hat{G}}(q) \quad \text{--- ②}$$

$\underline{\hat{M}}, \underline{\hat{C}}, \underline{\hat{G}}$ \rightarrow estimates of M, C, G

lets assume $M = \underline{\hat{M}}, C = \underline{\hat{C}}, G = \underline{\hat{G}}$

Substitute ② in ①

$$M \ddot{q} + C(q, \dot{q}) \dot{q} + G(q) = M (-k_p q - k_d \dot{q}) + C(q, \dot{q}) \dot{q} + G(q)$$

$$M (\ddot{q} + k_d \dot{q} + k_p q) = 0$$

$$M \neq 0 \quad \ddot{q} + k_d \dot{q} + k_p q = 0 \quad \leftarrow$$

$$\begin{bmatrix} \ddot{q}_1 \\ \ddot{q}_2 \\ \vdots \\ \ddot{q}_n \end{bmatrix} + \begin{bmatrix} k_{d1} & 0 & 0 & \dots & - \\ 0 & k_{d2} & 0 & \dots & - \\ \dots & \dots & \dots & \dots & \dots \\ 0 & 0 & \dots & - & k_{dn} \end{bmatrix} \begin{bmatrix} \dot{q}_1 \\ \dot{q}_2 \\ \vdots \\ \dot{q}_n \end{bmatrix} + \begin{bmatrix} k_{p1} & 0 & \dots & - \\ 0 & k_{p2} & \dots & - \\ \vdots & \vdots & \dots & \vdots \\ 0 & \dots & \dots & k_{pn} \end{bmatrix} \begin{bmatrix} q_1 \\ q_2 \\ \vdots \\ q_n \end{bmatrix} = 0$$

✓ ✓ ✓

These equations are decoupled

$$\rightarrow \ddot{q}_1 + \underline{k_{d1}} \dot{q}_1 + \underline{k_{p1}} q_1 = 0$$

$$\ddot{q}_2 + \underline{k_{d2}} \dot{q}_2 + \underline{k_{p2}} q_2 = 0$$

⋮

$$\ddot{q}_n + \underline{k_{dn}} \dot{q}_n + \underline{k_{pn}} q_n = 0$$

*n decoupled
equations*

NOTE: $\underline{m} \ddot{q} + (\underline{k_d} + \underline{c}) \dot{q} + (\underline{k_p} + \underline{k}) q = 0$

Critically damped: $\underline{k_d} = -\underline{c} + 2\sqrt{(\underline{k} + \underline{k_p})\underline{m}}$

↑ derived earlier

$$k_{d1} = 0 + 2\sqrt{(0 + k_{p1})m}$$

$$k_{d1} = 2\sqrt{k_{p1}m}$$

⋮

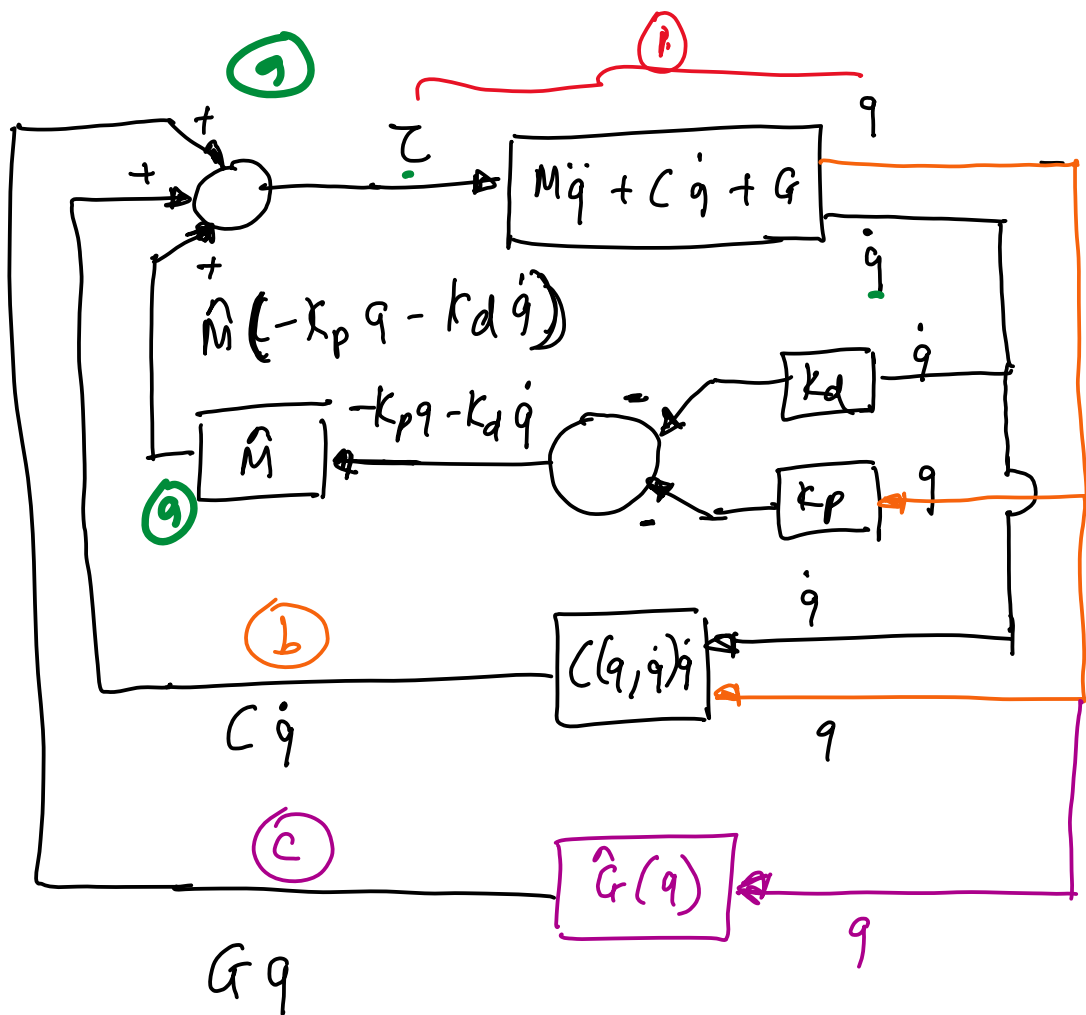
$$k_{dn} = 2\sqrt{k_{pn}m}$$

→ $k_{di} = 2\sqrt{k_{pi}m}$ ←

Block Diagram

$$\sqrt{M(q)\ddot{q} + C(q, \dot{q})\dot{q} + G(q) = \tau} \quad \text{--- (1)}$$

$$\sqrt{\tau = \hat{M}(-k_p q - k_d \dot{q}) + \hat{C}(q, \dot{q})\dot{q} + \hat{G}(q)} \quad \text{--- (2)}$$



$q, \dot{q} \rightarrow$ outputs (sensors)

$\tau \rightarrow$ input (actuator)

Code Example 1

Control - partitioning - pd

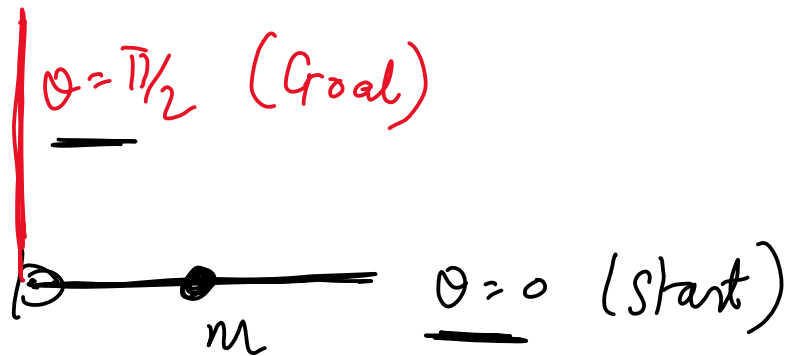
$$\begin{array}{ccccccc} \cdot M \dot{q} & + & C \dot{q} & + & G q & = & F \\ \downarrow & & \downarrow & & \downarrow & & \downarrow \\ 2 \times 2 & & 2 \times 1 & & 2 \times 2 & & 2 \times 1 \\ & & & & \downarrow & & \downarrow \\ & & & & 2 \times 1 & & 2 \times 1 \\ & & & & & & \downarrow \\ & & & & & & 2 \times 1 \end{array}$$

~~(1) $F = -k_p q - k_d \dot{q}$ (X)~~

(2) $F = \underline{M} (-k_p (q - q_{des}) - k_d \dot{q}) + \underline{C} \dot{q} + \underline{G} q$ ✓

(3) $F = \underline{\hat{M}} (-k_p (q - q_{des}) - k_d \dot{q}) + \underline{\hat{C}} \dot{q} + \underline{\hat{G}} q$

Example: 1-link pendulum



$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + G(q) = \tau$$

$$(I + ml^2)\ddot{\theta} + 0 + \underbrace{mgl \sin \theta}_{\text{gravity}} = \tau$$

Controllers

$$\checkmark \textcircled{1} \quad \tau = -k_p (q - q_{des}) - k_d \dot{q} \quad \checkmark$$

$$\checkmark \textcircled{2} \quad \tau = M(-k_p (q - q_{des}) - k_d \dot{q}) + C(q, \dot{q})\dot{q} + G(q)$$

\downarrow
 $(I + ml^2)$

\downarrow
 $(\pi/2)$

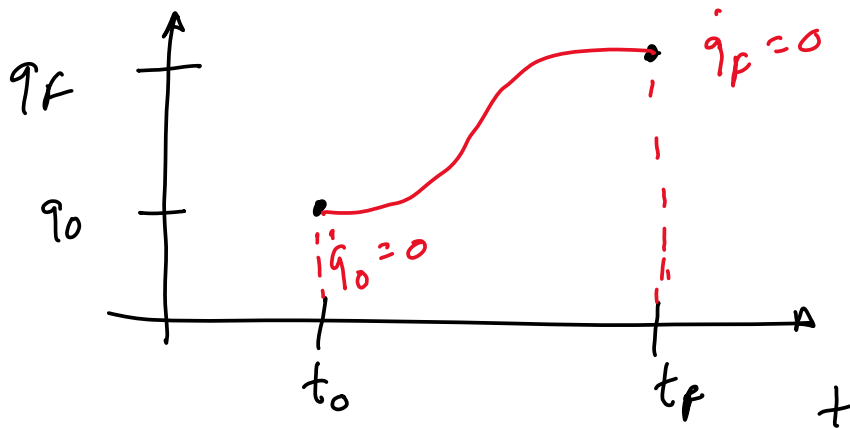
\parallel
 0

\parallel
 $mgl \sin \theta$

(3a - pendulum - pd) - This folder

(3b - pendulum - control - partitioning)

Trajectory tracking in joint space



$$\checkmark q_{ref}(t) = a_0 + a_1 t + a_2 t^2 + a_3 t^3 \quad \checkmark$$

$$\text{Dynamics} \quad M \ddot{q} + C\dot{q} + G = \tau \quad \text{--- (1)}$$

$$\text{Control:} \quad \tau = M(\ddot{q}_{ref} - k_d(\dot{q} - \dot{q}_{ref}) - k_p(q - q_{ref})) + G(q) + C(q, \dot{q})\dot{q} \quad \text{--- (2)}$$

Why does this τ work?

Substitute (2) in (1)

$$M\ddot{q} + \cancel{C\dot{q}} + \cancel{G} = M(\ddot{q}_{ref} - k_d(\dot{q} - \dot{q}_{ref}) - k_p(q - q_{ref})) + \cancel{C\dot{q}} + \cancel{G}$$

$$M\ddot{q} + \cancel{(\dot{q} + \hat{r})} = M(\dot{q}_{ref} - k_d(\hat{q} - \dot{q}_{ref})) - k_p(q - q_{ref}) + \cancel{(\dot{q} + \hat{r})}$$

$$M\ddot{q} = M(\dot{q}_{ref} - k_d(\hat{q} - \dot{q}_{ref}) - k_p(q - q_{ref}))$$

$$M((\ddot{q} - \ddot{q}_{ref}) + k_d(\dot{q} - \dot{q}_{ref}) + k_p(q - q_{ref})) = 0$$

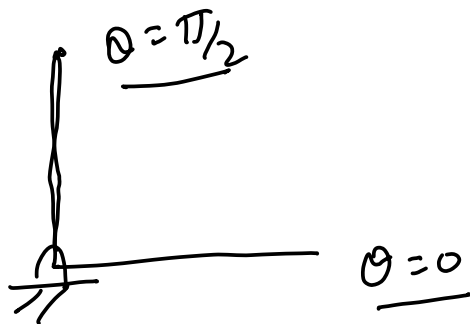
$$M \neq 0$$

$$(\ddot{q} - \ddot{q}_{ref}) + k_d(\dot{q} - \dot{q}_{ref}) + k_p(q - q_{ref}) = 0$$

$$e = q - q_{ref}$$

$$\rightarrow \underbrace{\ddot{e}}_{n \times 1} + k_d \underbrace{\dot{e}}_{n \times 1} + k_p \underbrace{e}_{n \times 1} = 0 \quad \text{Similar to spring-mass damper eqn.}$$

EXAMPLE: Trajectory Tracking of a 1-link pendulum



$$\theta = 0 \quad \rightarrow \quad \theta = \pi/2 \quad \theta_{1, \text{ref}} = a_{10} + a_{11}t + a_{12}t^2 + a_{13}t^3$$

$$t = 0 \quad \quad \quad t = t_1$$

$$\theta = \pi/2 \quad \rightarrow \quad \theta = 0 \quad \theta_{2, \text{ref}} = a_{20} + a_{21}t + a_{22}t^2 + a_{23}t^3$$

$$t = t_1 \quad \quad \quad t = t_2$$

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + G(q) = \tau$$

$$ml^2\ddot{\alpha} + 0 + mg \sin \theta = \tau$$

Controller

$$\tau = \underbrace{M}_{ml^2} (\ddot{\theta}_{\text{ref}} - k_p(\theta - \theta_{\text{ref}}) - k_d(\dot{\theta} - \dot{\theta}_{\text{ref}})) + \underbrace{C(q, \dot{q})\dot{q}}_0 + \underbrace{G(q)}_{mg \sin \theta}$$

< 4a - pendulum - trajectory - tracking > } PYTHON
 < 4b - double pendulum - trajectory - tracking > }

$$X = f(q)$$

$$\dot{X} = \frac{\partial f}{\partial q} \dot{q} \quad (\text{Chain rule})$$

$$\dot{X} = J \dot{q}$$

$$\dot{X}_{\text{ref}} = J \dot{q}_{\text{ref}}$$

$$\underline{\dot{q}_{\text{ref}}} = J^{-1} \dot{X}_{\text{ref}} = J^{-1} \begin{bmatrix} \dot{x}_{\text{ref}} \\ \dot{y}_{\text{ref}} \end{bmatrix} \quad \text{--- (I)}$$

$$\ddot{X} = \dot{J} \dot{q} + J \ddot{q}$$

$$\ddot{X}_{\text{ref}} = \dot{J} \dot{q}_{\text{ref}} + J \ddot{q}_{\text{ref}}$$

$$\checkmark \ddot{q}_{\text{ref}} = J^{-1} [\ddot{X}_{\text{ref}} - \dot{J} \dot{q}_{\text{ref}}] \quad \text{--- (II)}$$

From (I), (II), (III) we have computed q_{ref} , \dot{q}_{ref} , \ddot{q}_{ref} from X_{ref} , \dot{X}_{ref} , \ddot{X}_{ref}

$$\tau = M(\ddot{q}_{\text{ref}} - k_p(q - q_{\text{ref}}) - k_d(\dot{q} - \dot{q}_{\text{ref}}))$$

<5 - double pendulum - cartesian control> PYTHON