

Ka

## 6.09 Endurance Limit Modify Factors

$$S_e \neq S_e' \Rightarrow S_e \leq S_e'$$

$S_e$  - endurance of material in machines

$S_e'$  - lab tested endurance

$$S_e = k_a k_b k_c k_d k_e S_e'$$

- $k_a$  - surface factor
- $k_b$  - size factor
- $k_c$  - load factor
- $k_d$  - temperature factor
- $k_e$  - reliability factor

### a) Surface factor

Specimen is highly polished, but machine element is not.

$$k_a = a S_{ut}^b$$

$S_{ut}$  is the tensile strength,

$a, b$  constants from the table below

| <u>Surface Finish</u>      | <u>Factor a</u>                 |                                | <u>Exponent b</u> |
|----------------------------|---------------------------------|--------------------------------|-------------------|
|                            | <u><math>S_{ut}</math> kpsi</u> | <u><math>S_{ut}</math> MPa</u> |                   |
| ✓ Ground                   | 1.21                            | 1.38                           | -0.067            |
| → ✓ Machined or cold-drawn | 2.00                            | 3.04                           | -0.217            |
| ✓ Hot-rolled               | 11.0                            | 38.6                           | -0.650            |
| ✓ As-forged                | 12.7                            | 54.9                           | -0.758            |

Kb

b) Size factor,  $k_b$

Rotating round specimen

(i) Bending and torsion

$$\left. \begin{array}{l} \underbrace{k_b}_{\sim} = \begin{cases} 0.879 d^{-0.107} \\ 0.91 d^{-0.157} \end{cases} & \begin{array}{l} 0.11 < d \leq 2 \text{ in} \\ 2 < d \leq 10 \text{ in} \end{array} \quad \parallel \text{ ips} \\ \underbrace{k_b}_{\sim} = \begin{cases} 1.24 d^{-0.107} \\ 1.51 d^{-0.157} \end{cases} & \begin{array}{l} 2.79 < d \leq 51 \text{ mm} \\ 51 < d \leq 254 \text{ mm} \end{array} \quad \parallel \text{ SI} \end{array} \right\} (*)$$

(ii) Axial

$$\underbrace{k_b}_{\sim} = 1$$

Non-rotating round / non-round specimen.

- Critical volume theory. Volume of material experiencing 95% of maximum stress is critical. The goal here is to use critical diameter  $d_c$  obtained by equating:

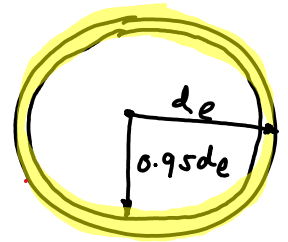
$$\begin{aligned} & \text{volume of 95\% stress of non-rotating specimen} \\ & = \text{volume of 95\% stress of rotating round specimen} \end{aligned}$$

When volumes are equated lengths drop out and we are left equating areas.

## Kb (contd)

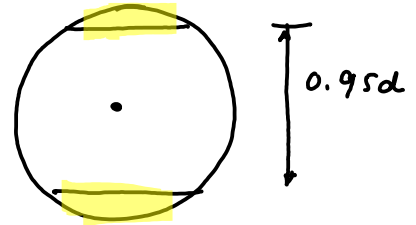
For rotating round specimen, the 95% area is

$$\star A_{0.95r} = \frac{\pi}{4} (d_e^2 - 0.95d_e^2) = 0.0766 d_e^2 \quad - (1)$$



For non-rotating round specimen

$$\star A_{0.95s} = 0.1046 d^2 \quad - (2)$$

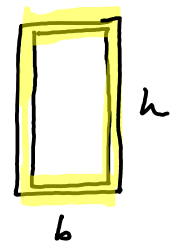


Equating (1) to (2)

$$\checkmark \boxed{d_e = 0.370 d}$$

For rectangular section

$$A_{0.95} = 0.05bh \quad - (3)$$

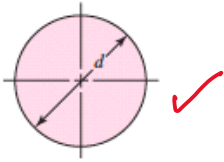


Equating (1) to (3)

$$\checkmark \boxed{d_e = 0.808 \sqrt{hb}}$$

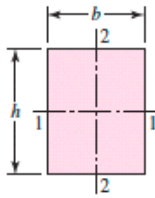
This  $d_e$  is used in place of  $d$  in equations  $\oplus$

$k_b$



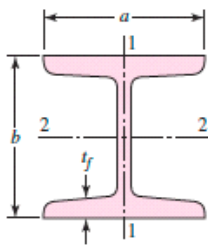
$$A_{0.95\sigma} = 0.01046d^2$$

$$d_e = 0.370d$$

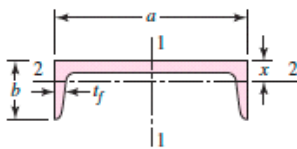


$$A_{0.95\sigma} = 0.05hb$$

$$d_e = 0.808\sqrt{hb}$$



$$A_{0.95\sigma} = \begin{cases} 0.10at_f & \text{axis 1-1} \\ 0.05ba & \text{axis 2-2} \end{cases} \quad t_f > 0.025a$$



$$A_{0.95\sigma} = \begin{cases} 0.05ab & \text{axis 1-1} \\ 0.052xa + 0.1t_f(b-x) & \text{axis 2-2} \end{cases}$$

(c) Loading factor ( $k_c$ )

$$k_c = \begin{cases} 1 & \text{bending} \\ 0.85 & \text{axial} \\ 0.59 & \text{torsion} \end{cases}$$

Kd

(d) Temperature factor (Kd)

$$\left. \begin{aligned} \frac{S_T}{S_{RT}} &= 0.98 + 3.5 (10^{-4}) T_F - 6.3 (10^{-7}) T_F^2 \\ \frac{S_T}{S_{RT}} &= 0.99 + 5.9 (10^{-4}) T_C - 2.1 (10^{-6}) T_C^2 \end{aligned} \right\} \textcircled{2}$$

$T_F$  - Fahrenheit ;

$T_C$  - Celsius

$S_T$  - ultimate strength at operating temperature

$S_{RT}$  - ultimate strength at room temperature

field  $\checkmark$  lab expt.

(i)  $S_e$  is unknown

$\checkmark$  (a)  $\checkmark$  Estimate  $S_T/S_{RT}$  from  $\textcircled{2}$

$\checkmark$  (b) Estimate  $(S_{ut})_T$  using

$$(S_{ut})_T = \left( \frac{S_T}{S_{RT}} \right) (S_{ut})_{RT}$$

Room temp

$\checkmark$  (c) Estimate  $S_e$  using

$$S_e' = \begin{cases} 0.5 (S_{ut})_T & (S_{ut})_T \leq 200 \text{ kpsi} \\ 100 \text{ kpsi} & (S_{ut})_T > 200 \text{ kpsi} \end{cases} \quad \parallel \text{ ips}$$

$$S_e' = \begin{cases} 0.5 (S_{ut})_T & (S_{ut})_T \leq 1400 \text{ MPa} \\ 700 \text{ MPa} & (S_{ut})_T > 1400 \text{ MPa} \end{cases} \quad \parallel \text{ SI}$$

(d) and  $K_d = 1$   $S_e = S_e'$

# Ke

(ii) Se' is known

- ✓ (a) Estimate  $\frac{S_T}{S_{RT}}$  from  $\hat{\sigma}$
- ✓ (b) Estimate  $k_d = S_T/S_{RT}$
- ✓ (c) Estimate  $S_e = k_d S_e'$

(e) Reliability factor (ke)

$$k_e = 1 - 0.08 z_a$$

| Reliability, % | Transformation Variate $z_a$ | Reliability Factor $k_e$ |
|----------------|------------------------------|--------------------------|
| 50             | 0                            | 1.000                    |
| 90             | 1.288                        | 0.897                    |
| 95             | 1.645                        | 0.868                    |
| 99             | 2.326                        | 0.814                    |
| 99.9           | 3.091                        | 0.753                    |
| 99.99          | 3.719                        | 0.702                    |
| 99.999         | 4.265                        | 0.659                    |
| 99.9999        | 4.753                        | 0.620                    |

$$S_e = k_e S_e'$$

Q1

A steel specimen has a minimum ultimate strength of 520 MPa has a machine surface. Estimate the factor  $K_a$

From the table, Row 2

$$a = 3.04$$

$$b = -0.217$$

$$S_{ut} = 520 \text{ MPa}$$

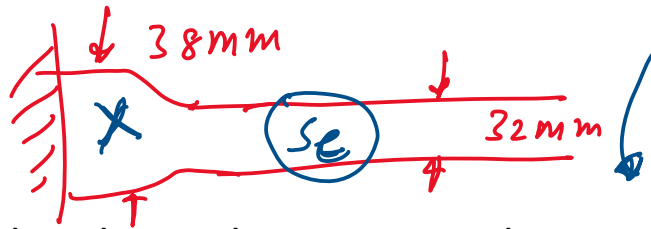
$$K_a = a S_{ut}^b$$

$$K_a = (3.04)(520)^{-0.217}$$

$$K_a = 0.78$$

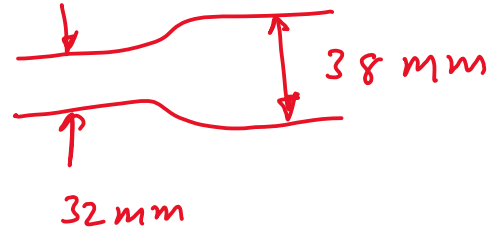
$$S_e = K_a S_e'$$

Q2



A steel shaft is loaded in bending. It has a 32 mm diameter and a filleted shoulder of 38 mm diameter. Estimate the size factor ( $k_b$ ) for

- (a) rotating mode  
 (b) nonrotating mode



(a) rotating mode

$$k_b = \begin{cases} 1.24 d^{-0.107} & 2.79 < d \leq 51 \text{ mm} \\ 1.51 d^{-0.157} & 51 < d \leq 254 \text{ mm} \end{cases} \quad \text{SI}$$

$d = 32 \text{ mm}$

$k_b = (1.24)(32)^{-0.107}$

$k_b = 0.855$  ✓

~~$d = 38 \text{ mm}$~~   
 ~~$k_b = (1.24)(38)^{-0.107}$~~

~~$k_b = 0.84$~~  ✓

~~$S_e = 0.84$~~  ✓

$S_e = (k_b) S_e'$   
 $0.84$

$S_e = 0.855$

NOTES

(b) non-rotating mode

$d_e = 0.37d = 0.37(32) \Rightarrow d_e = 11.84 \text{ mm}$

$k_b = (1.24) d_e^{-0.107} = (1.24)(11.84)^{-0.107}$



$\sigma_b = (1.24) \sigma_e$

$\sigma_b = (1.24) (11.04)$

$$K_b = 0.95$$

### Q3

A 1035 steel has a tensile strength of  $80 \text{ kpsi}$  and is to be used for a part that operates at a temperature of  $750 \text{ F}$ . Estimate the endurance limit at the operating temperature if

$S_e = ?$

(a) only the tensile strength at room temperature is known

(b) at the room-temperature of  $70 \text{ F}$ , the endurance limit for the material is found by test to be  $S_e' = 39 \text{ kpsi}$

$$(a) (S_{ut})_{RT} = (S_{ut})_{70F} = 80 \text{ kpsi} \quad RT = \text{room temp}$$

$$(S_e)_T = (S_e)_{750F} = ? \quad T = \text{operating temp}$$

$S_e'$  is unknown

(a)

$$\frac{S_T}{S_{RT}} = 0.98 + 3.5 (10^{-4}) T_F - 6.3 (10^{-7}) T_F^2$$

From  
kd  
data

$$\frac{S_T}{S_{RT}} = 0.98 + 3.5 (10^{-4}) (750) - 6.3 (10^{-7}) (750)^2$$

$$\frac{S_T}{S_{RT}} = \frac{S_{750}}{S_{70}} = 0.89$$

$$(b) (S_{ut})_T = \left( \frac{S_T}{S_{RT}} \right) (S_{ut})_{RT}$$

$$(S_{ut})_{750} = (0.89) (80) \Rightarrow (S_{ut})_{750} = 71.2 \text{ kpsi}$$

(c)

$$S_e' = \begin{cases} 0.5 (S_{ut})_T & (S_{ut})_T \leq 200 \text{ kpsi} \\ 100 \text{ kpsi} & (S_{ut})_T > 200 \text{ kpsi} \end{cases}$$

$$(S_{ut})_{750} = 71.2 \text{ kpsi} \leq 200 \text{ kpsi}$$

$$(S_e')_T = 0.5 (S_{ut})_T = 0.5 (S_{ut})_{750} = 0.5 (71.2)$$

$$(S_e')_{750} = 35.6 \text{ kpsi}$$

(d)  $k_d = 1$

$$S_e = S_e' = 35.6 \text{ kpsi}$$

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(ii)  $(S_e')_{RT} = (S_e')_{70} = \underline{39} \text{ kpsi}$

$S_e'$  is known

(a)  $S_T / S_{RT} = S_{750} / S_{70} = 0.89$  [see part (i)]

(b)  $k_d = S_T / S_{RT} = 0.89$

(c)  $(S_e)_{750} = k_d (S_e')_{70} = 0.89 (39) = 35 \text{ kpsi}$

$$(S_e)_{750} = 35 \text{ kpsi}$$

# Q4

A 1080 hot-rolled steel bar has been machined to a diameter of 1 in. It is to be placed in reversed axial loading for 70,000 cycles to failure in an operating environment of 650 F. Estimate the endurance limit and fatigue strength at 70,000 cycles. Assume reliability for the endurance limit estimate to be 99% and ultimate tensile strength ( $S_{ut}$ ) at 70 F to be 112 kpsi.

$$d = 1 \text{ in} \quad ; \quad N = 70,000 \quad ; \quad RT = 70 \text{ F} \quad ; \quad T = 650 \text{ F}$$

$$R_e = 99\% \quad ; \quad (S_{ut})_{RT} = (S_{ut})_{70} = 112 \text{ kpsi}$$

$$(i) \quad S_e = ? \quad (ii) \quad S_f = a N^b$$

$$S_e = k_a k_b k_c k_d k_e S_e' \quad || \quad a = \frac{(f S_{ut})^2}{S_e} \quad ; \quad b = \frac{1}{3} \log \left( \frac{S_e}{f S_{ut}} \right)$$

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$$(i) \quad k_a = a (S_{ut})^b = 11 (?)^b$$

$$(S_{ut})_T = (S_{ut})_{650}$$

(i)

$$\frac{S_T}{S_{RT}} = 0.98 + 3.5 (10^{-4}) T_F - 6.3 (10^{-7}) T_F^2$$

from Ka

(a)  $\frac{S_T}{S_{RT}} = 0.98 + 3.5 (10^{-4}) (650) - 6.3 (10^{-7}) (650)^2$

$$\frac{S_{650}}{S_{70}} = 0.94$$

(b)  $(S_{ut})_T = \left( \frac{S_T}{S_{RT}} \right) (S_{ut})_{RT}$

$$(S_{ut})_{650} = (0.94) (112) = 105 \text{ kpsi} \quad \text{--- (I)}$$

(c)

$$S_e' = \begin{cases} 0.5 (S_{ut})_T & (S_{ut})_T \leq 200 \text{ kpsi} \\ 100 \text{ kpsi} & (S_{ut})_T > 200 \text{ kpsi} \end{cases} \quad \text{|| ips}$$

Copied from Ka

$$S_e' = (0.5) (S_{ut})_{650} = (0.5) (105)$$

$$S_e' = 52.5 \text{ kpsi}$$

(d)

$$\underline{K_d = 1}$$

$$S_e = \underline{S_e'} = \underline{52.5 \text{ kpsi}}$$

$$(i) \quad k_a = a (\underline{S_{ut}})^b = 11 (?)^b$$

$\underbrace{\hspace{10em}}_{(S_{ut})_T = (S_{ut})_{650}}$

$$= 11 (105)^{-0.65}$$

$$k_a = 0.534$$

$$k_b = 1 \quad (\text{axial})$$

$$k_c = 0.85 \quad (\text{axial})$$

$$k_d = 1$$

$$k_e = 0.814 \quad (99\% \text{ reliability})$$

$$S_e = (k_a)(k_b)(k_c)(k_d)(k_e) S_e'$$

$$= (0.534)(1)(0.85)(1)(0.814)(52.5)$$

↑ at  $T = 650^\circ\text{F}$

$$S_e = 19.39$$

$$(ii) \quad a = \frac{(f S_{ut})^2}{S_e} \quad b = \frac{1}{3} \log \left( \frac{S_e}{f S_{ut}} \right)$$

$$(S_{ut})_T = (S_{ut})_{650} = \underline{105} \text{ kpsi}$$

$$\text{At } S_{ut} = 105 \text{ kpsi} \quad f = 0.84 \text{ (see Lec 17)}$$

$$a = \frac{[(0.84)(105)]^2}{19.39} = 401.2$$

$$b = \frac{1}{3} \log \left( \frac{19.39}{0.84(105)} \right) = -0.22$$

$$S_f = a N^b = (401.2) (70,000)^{-0.22}$$

$$S_f = 34.47 \text{ kpsi}$$

